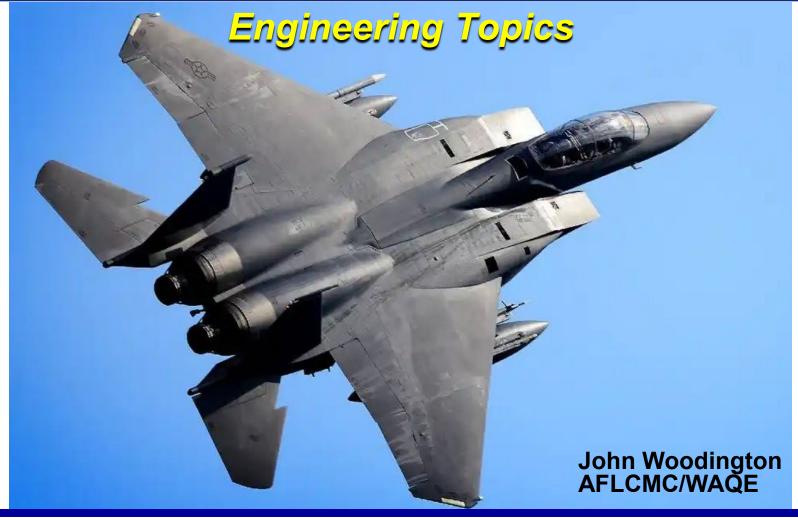


2024 F-15 Worldwide Review







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FDO Case # WR-24-744-XX



Overview



- F-15E Longeron Repair Initiative
- IAT 49: Front Spar Tooling Hole Cracks
- Certified Service Life Re-Certification for F-15C/D and F-15E



F-15E Longeron Repair Initiative



U.S. AIR FORCE

- F-15E aircraft will receive a planned Longeron Repair Initiative (LRI) involving the replacement of the main longerons due to the complexity and burden of required inspections
 - Approximately 1700 manhours to accomplish
- Holes in new longerons will be 100% coldworked and corrosion preventatives will be applied
 - This helps to ensure maximum life of the parts and reduce future inspections
- FTE-10 did reach the end of the test with the original longerons
 - LRI is being accomplished to reduce the maintenance and engineering burden of aging safety of flight parts and increase aircraft availability
- Parts will be procured through DLA
- First aircraft successfully completed LRI Oct 2024

	QTY	NSN	Part Number	Name
	1	1560-01-600-0804	68A313171-2001	Canopy Sill Longeron
	1	1560-01-600-0803	68A313171-2002	Canopy Sill Longeron
	1	1560-01-620-6730	68A321024-2063	Upper Inboard Longeron
	1	1560-01-620-6724	68A321024-2060	Upper Inboard Longeron
	1	1560-01-696-1626	68A324501-2011A	IAT 6 bracket
	1	1560-01-696-1624	68A324501-2012A	IAT 6 bracket
	1	1560-01-696-1428	68A324501-2009A	IAT 6 bracket
	1	1560-01-697-5562	68A324501-2010A	IAT 6 bracket
	1	1560-01-706-8000	68A321085-2009	IAT 6 bracket
	4	1560-01-696-1440	68A321085-2007	IAT 6 bracket
	1	1560-01-633-9979	68A316075-2007A	IAT 69 bracket
	1	1560-01-634-0307	68A316075-2008A	IAT 69 bracket
	1	1560-01-425-4076	68A318024-2049	Skin
	1	1560-01-425-4075	68A318024-2050	Skin
	1	1560-01-425-4079	68A318024-2053	Skin
	1	1560-01-425-4080	68A318024-2054	Skin
	1	1560-01-673-0775	68A350605-2003	Windscreen fairing
	1	1560-01-673-0778	68A350605-2004	Windscreen fairing
	1	1560-01-711-1143	201873455-01	Repair Bracket
	1	1560-01-711-1141	201873455-02	Repair Bracket
	1	1560-01-628-3989	68A318015-2031	Longeron clip
	1	1560-01-628-3995	68A318015-2032	Longeron clip
	24	5340-01-535-4789	FTR-3-0-1250-S	Windscreen Force-Tec Retainers
	1	1560-01-714-1474	68A321309-2003	Beam (FWD)
	1	1560-01-714-1490	68A321309-2004	Beam (FWD)
	1	1560-01-714-1266	68A321028-2029	Angle
	1	1560-01-714-1271	68A321028-2030	Angle
	1	1560-01-714-1279	68A321028-2051	Stringer
	1	1560-01-714-1285	68A321028-2052	Stringer
_		(OT II)		



IAT 49 Front Spar Tooling Hole Cracks



- IAT 49.0 Outboard Front Spar Tooling Holes
 - 10" crack found at IAT 49.0 during unrelated field level inspection prompted installation of outboard front spar splice
 - TCTO released in Dec 2022 to inspect forward side of front spar at IAT 49.0
- Repairs Approaches:
 - Large cracks into upper/lower flanges: Outboard front spar splice installed or spar replacement
 - Smaller cracks: Temporary double L-Angle repair installed by field units with front spar splice installed at depot
- Between Dec 2022 and Oct 2024, 240 wings have been inspected
 - 45 wings had at least one crack indication
 - In the same period, configuration updates show 72% of wings already had or just received a front spar splice/new front spar







Certified Service Life for F-15C/D



- Engineering has submitted an airworthiness package to extend the service life of F-15C/D fleet to ½ of the Test Demonstrated Life (TDL), 14016.5 FTA-7 EFH
 - The timeframe for completion is no later than FY25
 - The goal is to re-certify the fleet prior to needing a new risk acceptance for continued flight
 - Known mitigating actions required for recertification
 - Canopy sill longeron, Upper aft longeron, and Upper inboard longeron must be replaced
 - All IAT inspections must continue to be done
 - Additional mitigating actions may be required





Certified Service Life for F-15E



- Engineering has recertified the USAF F-15E fleet to a new service life to 1/3 of the TDL, 16,332 FTE-10 EFH
 - Interim extension utilized a risk-based approach that incorporated USAF F-15E fleet usage, IAT inspection results, updated DTA analysis, and FTE-10 "in-test" findings as the basis for recertification
 - Mechanical components and avionics were also addressed in the recertification but did not require any mitigating actions
 - USAF F-15E aircraft required structural mitigating actions to be certified to the new 16,332 life
 - All IAT's must continue to be performed
 - All holes in IAT 69 must be coldworked IAW 1F-15-1679





Certified Service Life for F-15E



- Recertification to 1/2 TDL, 24,498 FTE-10 EFH is in work but may be several years before completed
- FTE-10 Teardown has been completed by S&K and all findings have been sent to Boeing
 - Boeing is using these findings to create or update DTA's
 - USAF will then use these new DTA's to form updated inspection criteria and/or PDM tasks
- Recertification to 24,498 may have additional mitigating requirements that must be complied with to take advantage of the new certified life
 - These mitigating actions are unknown at this time and will be established during the recertification process



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